

Rotor Design Approaches

Michael S. Selig
Associate Professor

Department of Aerospace Engineering
University of Illinois at Urbana-Champaign

Steady-State Aerodynamics Codes for HAWTs
Selig, Tangler, and Giguère



University of Illinois at Urbana-Champaign
National Renewable Energy Laboratory



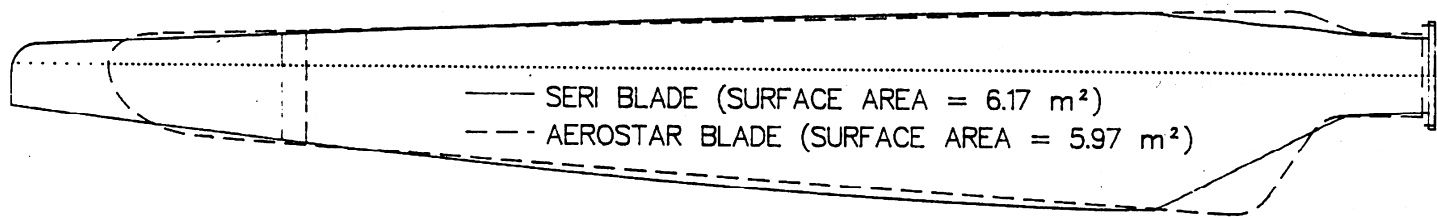
Outline

- Design Problems
- Approaches to Design
- Inverse Design

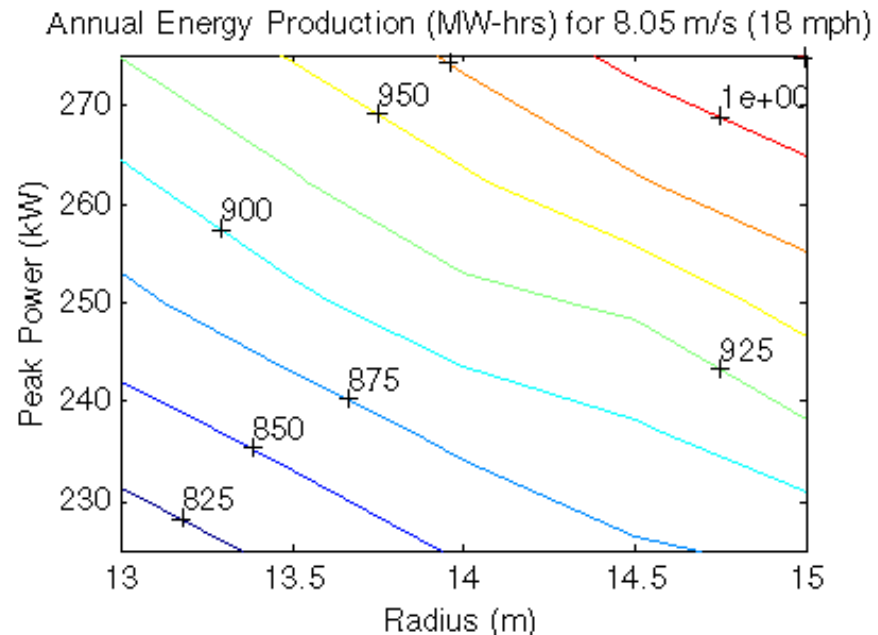


Design Problems

- Retrofit Blades

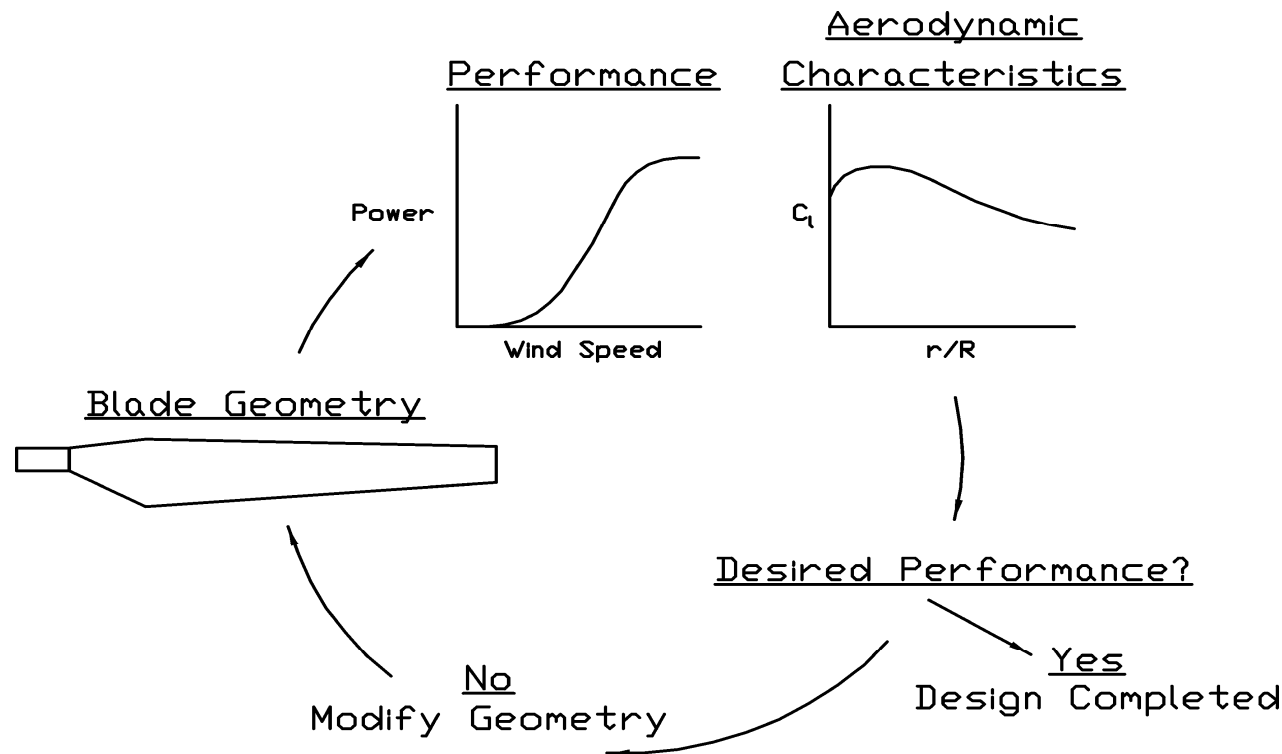


- Trade Studies and Optimization



Approaches to Design

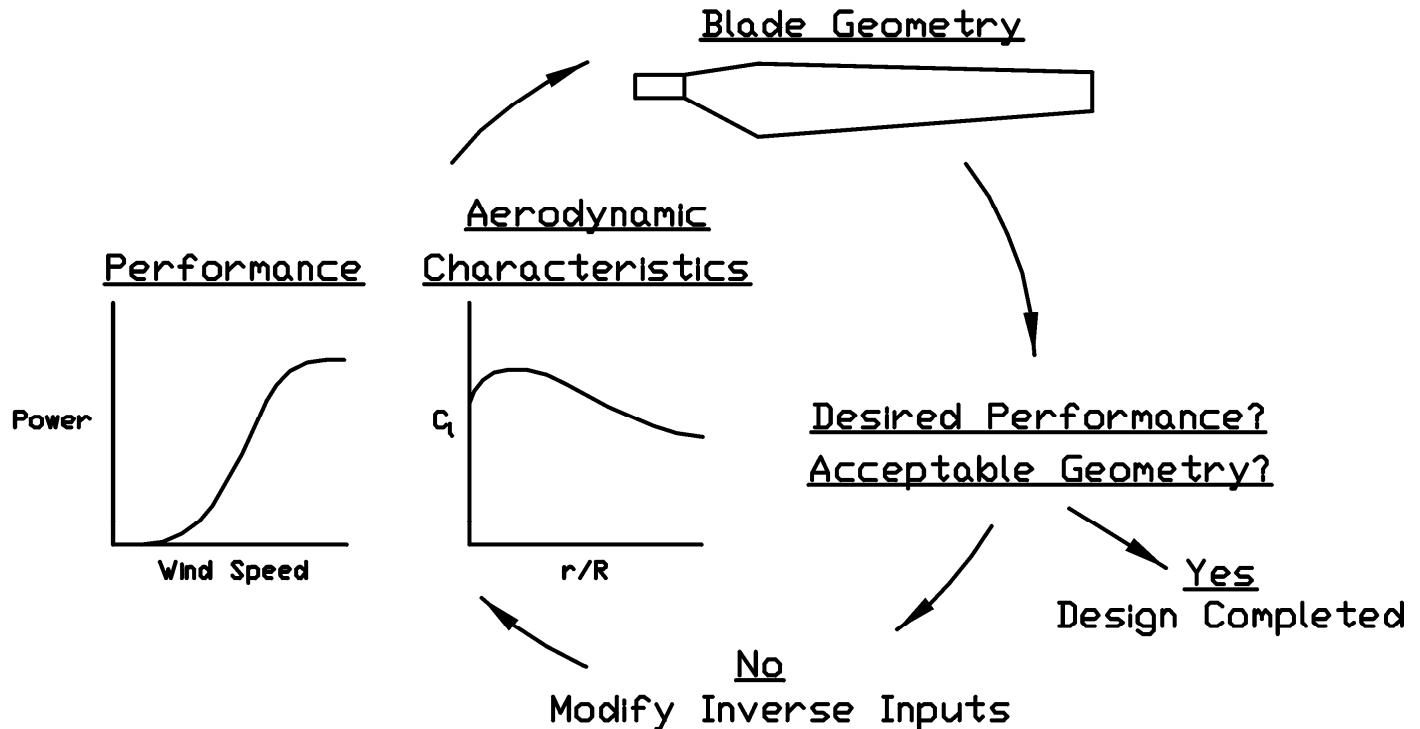
- Design by Analysis - Working “Forward”



- Example HAWT Codes: PROP, PROP93, WT_PERF, PROPID



- Inverse Design - Working “Backward”
 - Requires some knowledge of desirable aerodynamic characteristics



- Example HAWT Code: PROPID



- Optimization
 - Example problem statement: Maximize the annual energy production subject to various constraints given a set of design variables for iteration
- Example Code: PROPGA

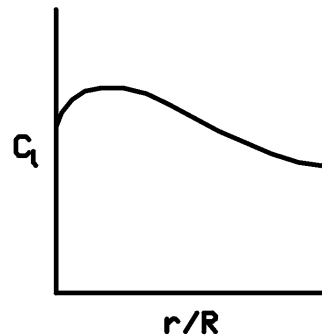


Design Variables

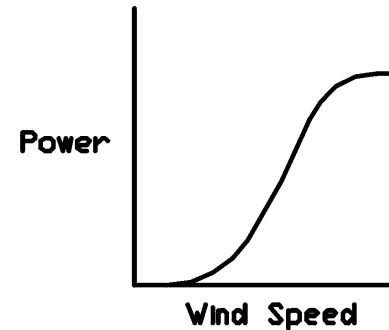
Blade Geometry



Aerodynamic Characteristics



Performance



- Desire a method that allows for the specification of both independent and dependent variables



From an Inverse Design Perspective

- Through an inverse design approach (e.g., PROPID), blade performance characteristics can be prescribed so long as associated input variables are given up for iteration.
- Examples:
 - Iterate on:
 - Rotor radius
 - Twist distribution
 - Chord distribution
 - To achieve prescribed:
 - => Rotor peak power
 - => Lift coef distribution
 - => Axial inflow distribution



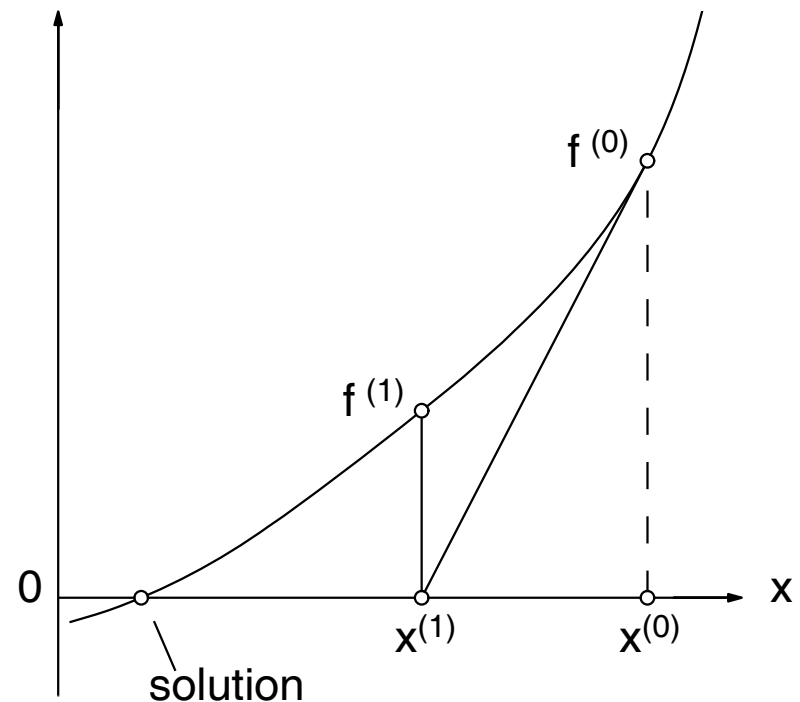
- Caveats
 - Some knowledge of the interdependence between the variables is required, e.g. the connection between the twist distribution and lift coefficient distribution.
 - Not all inverse specifications are physically realizable. For example, a specified peak power of 1 MW is not consistent with a rotor having a 3-ft radius.



Iteration Scheme

- Multidimensional Newton iteration is used to achieve the prescribed rotor performance
- Special "Tricks"
 - Step limits can be set to avoid divergence
 - Iteration can be performed in stages
 - Parameterization of the input allows for better convergence

$$f = P_{current} - P_{desired}$$



PROPID for Analysis

- Traditional Independent Variables (Input)
 - Number of Blades
 - Radius
 - Hub Cutout
 - Chord Distribution
 - Twist Distribution
 - Blade Pitch
 - Rotor Rotation Speed (rpm) or Tip-Speed Ratio
 - Wind Speed
 - Airfoils



- Traditional Dependent Variables - Sample (Output)
 - Power Curve
 - Power Coefficient C_p Curve
 - Rated Power
 - Maximum Power Coefficient
 - Maximum Torque
 - Lift Coefficient Distribution
 - Axial Inflow Distribution
 - Blade L/D Distribution
 - Annual Energy Production
 - Etc
- Each of these dependent variables can also be prescribed using the inverse capabilities of PROPID



Aerodynamic Design Considerations

- Clean vs. Rough Blade Performance
- Stall Regulated vs. Variable Speed
- Fixed Pitch vs. Variable Pitch
- Site Conditions, e.g. Avg. Wind Speed and Turbulence
- Generator Characteristics, e.g., Small Turbines, Two Speed



Common Design Drivers

- Generator => Peak Power
- Gearbox => Max Torque
- Noise => Tip Speed
- Structures => Airfoil Thickness
- Materials => Geometric Constraints
- Cost

